

Helicopter Technology Company

Maintenance Manual

HTCM-006

Part Number 204P2100-101 and -103 Models UH-1H, UH-1B*, TH-1F, UH-1F, UH-1P, 204B, 205A, and 205A-1 MAIN ROTOR BLADE (Installation and Maintenance)

* with STC # SR00026DE Installed

Initial Release Date: 12/17/2013

Revision		
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Change Date:	01/03/2020	

D	Added -103 Configuration	01/03/2020	GHB
С	Added Applic. – TH-1F, UH-1F, UH-1P, 204B, 205A, 205A-1	05/ 12/ 2015	GHB
В	Added Applicability - UH-1B with STC #SR00026DE Installed	04/ 10/ 2015	GHB
Α	Added Applicability - All FAA Certificated UH-1H	03/ 31/ 2014	GHB
N/C	Released	12/ 17/ 2013	GHB
	Note		
See S	ection "REFERENCE DOCUMENTS" below for indica	ted reference	ces.
The r	nost current revision of this document (HTCM-006) wi	ill be availa	ble
on th	e Helicopter Technology Company (HTC) website at		
www	<u>.helicoptertech.com</u> under Technical Publications.		
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 HELICOPTER TECHNOLOGY COMPANY MAINTENANCE MANUAL

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PURPOSE

The purpose of this document is to both provide instructions for Maintenance and Use of HTC produced Main Rotor Blade Assemblies and to provide an index to the relevant Chapters or Sections of the applicable Maintenance Manual for the particular model of helicopter.

It is extremely important that these relevant sections of the be followed precisely.

The most current revision of this document (HTCM-006) will be available on the Helicopter Technology Company (HTC) website at <u>www.helicoptertech.com</u> under Technical Publications.

- **Note:** As applicable and unless otherwise noted, all references below are to be found in:
 - Technical Manual –

Aviation and Intermediate Maintenance Instructions Army Model UH-1H/V/EH-1H/X (TM 55-1520-210-23-1).

• Helicopter Maintenance Manual -

Supplement No. 1 to Army Model UH-1B Helicopter (Restricted Category).

• Technical Manual –

Organizational Maintenance USAF Models TH-1F, UH-1F, and UH-1P Helicopters - T.O. 1H-1(U)F-2-1.

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• Maintenance and Overhaul Instructions -

Bell Model 204B – BHT-204B-M&O-1.

• Maintenance Manual -

Bell Model 205A-1 - BHT-205A1-MM-1.

Note: HTC has used the best possible materials for the construction of its Main Rotor Blade Assemblies.

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Helicopter Technology Company Maintenance Manual HTCM-006 UH-1H MAIN ROTOR BLADE

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REFERENCE DOCUMENTS

- Technical Manual Aviation and Intermediate Maintenance Instructions Army Model UH-1H/V/EH-1H/X (TM 55-1520-210-23-1), Headquarters, Department of the Army, Washington, D.C., Change 47, dated 20 September 2005, or later approved revision.
- Technical Manual Preventative Maintenance Daily Inspection Checklist (TM 55-1520-210-PMD), Headquarters, Department of the Army, Washington, D.C.
- Technical Manual Phased Maintenance Checklist (TM 55-1520-210-PM), Headquarters, Department of the Army, Washington, D.C.
- Federal Aviation Administration (FAA) Supplemental Type Certificate (STC) Number SR00026DE, Rotorcraft Development Corp., Corvalis, Montana, dated 06 September 2012, or later approved revision.
- Helicopter Maintenance Manual Supplement No. 1 to Army Model UH-1B Helicopter (Restricted Category), San Joaquin Helicopters, Delano, California.
- Technical Manual Organizational Maintenance USAF Models TH-1F, UH-1F, and UH-1P Helicopters - T.O. 1H-1(U)F-2-1, Secretary of the Air Force, Washington, D.C., Change 14, dated 28 April 1988, or later approved revision.
- Technical Manual Scheduled Inspection and Maintenance Requirements USAF Series TH-1F, UH-1F, and UH-1P Helicopters -T.O. 1H-1(U)F-2-6, Secretary of the Air Force, Washington, D.C., Change 16, dated 16 February 1988, or later approved revision.

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HELICOPTER TECHNOLOGY COMPANY MAINTENANCE MANUAL HTCM-006 UH-1H MAIN ROTOR BLADE Change Letter: D Change Date: 01/03/2020

- Maintenance and Overhaul Instructions Bell Model 204B BHT-204B-M&O-1, Bell Helicopter, a Textron Company, Fort Worth, Texas, Revision 10, dated 24 June 2011, or later approved revision.
- Maintenance Manual Bell Model 205A-1 BHT-205A1-MM-1, Bell Helicopter, a Textron Company, Fort Worth, Texas, Revision 7, dated 08 July 2013, or later approved revision.
- FAA Airworthiness Directive (AD) 2018-02-07 Applicability: TH-1F, UH-1B with STC SR0026DE Installed, UH-1F,UH-1H, and UH-1P, dated 01 February 2018
- FAA Airworthiness Directive (AD) 2018-02-08 Applicability: 204B,
 205A, and 205A-1, dated 01 February 2018
- Mandatory Service Bulletin (SB) 204-2100-1R3, Main Rotor Blade Root End Periodic Inspection and Protection, Revision 3 – Applicability: UH-1H, UH-1B with STC SR0026DE Installed, TH-1F, UH-1F, UH-1P, 204B, 205A, 205A-1, Helicopter Technology Company (HTC), dated 12 December 2017.

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DEFINITIONS AND ABBREVIATIONS

As applicable and unless otherwise noted, all Definitions and Abbreviations are to be found in:

• Technical Manual –

Aviation and Intermediate Maintenance Instructions Army Model UH-1H/V/EH-1H/X (TM 55-1520-210-23-1).

- Helicopter Maintenance Manual Supplement No. 1 to Army Model UH-1B Helicopter (Restricted Category).
- Technical Manual Organizational Maintenance USAF Models TH-1F, UH-1F, and UH-1P Helicopters - T.O. 1H-1(U)F-2-1.
- Maintenance and Overhaul Instructions –
 Bell Model 204B BHT-204B-M&O-1.
- Maintenance Manual –

Bell Model 205A-1 – BHT-205A1-MM-1.

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WEIGHT AND BALANCE

Installation of the Helicopter Technology Company (HTC) Part Number **204P2100-101 and/or -103** Main Rotor Blade is a direct replacement the Bell Helicopter - Textron Part Number **204-011-250-113** Main Rotor Blade and does not constitute a change to the Weight and Balance of the aircraft.

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WARNINGS

As applicable and unless otherwise noted, all Warnings are to be found in:

• Technical Manual –

Aviation and Intermediate Maintenance Instructions Army Model

UH-1H/V/EH-1H/X (TM 55-1520-210-23-1).

- Helicopter Maintenance Manual Supplement No. 1 to Army Model UH-1B Helicopter (Restricted Category).
- Technical Manual –

•

Organizational Maintenance USAF Models TH-1F, UH-1F, and UH-1P Helicopters - T.O. 1H-1(U)F-2-1.

- Maintenance and Overhaul Instructions Bell Model 204B – BHT-204B-M&O-1.
 - Maintenance Manual Bell Model 205A-1 – BHT-205A1-MM-1.

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HELICOPTER TECHNOLOGY COMPANY MAINTENANCE MANUAL HTCM-006 UH-1H MAIN ROTOR BLADE Change Letter: D Change Date: 01/03/2020

INDEX

Index is to be found in the Technical Manual – Aviation and Intermediate Maintenance Instructions Army Model UH-1H/V/EH-1H/X (TM 55-1520-210-23-1), Headquarters, Department of the Army, Washington D.C. unless otherwise noted.

As applicable and unless otherwise noted, the Index is to be found in:

• Technical Manual –

Aviation and Intermediate Maintenance Instructions Army Model UH-1H/V/EH-1H/X (TM 55-1520-210-23-1).

- Helicopter Maintenance Manual Supplement No. 1 to Army Model UH-1B Helicopter (Restricted Category).
- Technical Manual –

Organizational Maintenance USAF Models TH-1F, UH-1F, and UH-1P Helicopters - T.O. 1H-1(U)F-2-1.

- Maintenance and Overhaul Instructions –
 Bell Model 204B BHT-204B-M&O-1.
- Maintenance Manual –

Bell Model 205A-1 – BHT-205A1-MM-1.

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HTCM	-006	UH-1H MAIN ROTOR BLADE	
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AIRWORTHINESS LIMITATIONS

The Airworthiness Limitations section is FAA Approved and specifies maintenance required under §§43.16 and 91.403 of the Federal Aviation Regulations unless an alternative program has been FAA Approved.

Revision	Date	FAA Signature and Date
D	4/13/2020	
	N. SATING THE STATE OF THE STATE	
С	10/08/2015	Maureen Moreland (on File)
В	06/08/2015	Greg DiLibero (on File)
Α	06/17/2014	Ronald Atmur (on File)
N/C	01/13/2014	Greg DiLibero (on File)

Interchangeability and Life Limit

204P2100-101:

The Helicopter Technology Company (HTC) Part Number **204P2100-101** Main Rotor Blade is a direct replacement the Bell Helicopter - Textron Part Number **204-011-250-113** Main Rotor Blade. The HTC Main Rotor Blade is **fully interchangeable** and carries a **life-limit of 2,600 hours**.

Installation of this Blade is covered by FAA Supplemental Type Certificate (STC) Number SR02492LA.

Note: Review FAA Airworthiness Directive (AD) 2018-18-07 and FAA Airworthiness Directive (AD) 2018-18-08 to determine applicability. Perform Inspections at the required Intervals.

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HTCM	-006 UH-1H	MAIN ROTOR BLADE	
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204P2100-103:

The Helicopter Technology Company (HTC) Part Number **204P2100-103** Main Rotor Blade is a direct replacement the Bell Helicopter - Textron Part Number **204-011-250-113** Main Rotor Blade. The HTC Main Rotor Blade is **fully interchangeable** and carries a **life-limit of 2,600 hours**.

The -103 blade features a solid Titanium Lower Grip Plate.

Installation of this Blade is covered by FAA Supplemental Type Certificate (STC) Number **SR02492LA**.

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HTCM-006 UH-1H MAIN ROTOR BLADE				
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Part Number 204P2100-101 and -103 Main Rotor Blade is applicable on the following UH-1H Models:

JJASPP Engineering Services (R00004RC) Tamarack Helicopters (R00010SE) OAS Parts LLC (H7SO) Rotorcraft Dev. Corp. (H13WE) Southwest Florida Aviation (H6SO) Global Helicopter Technology (R00002RC) Hagglund Helicopters (H15NM) Arrow Falcon Exporters (R00007DE) Richard's Heavylift Helo Inc. (H3SO) Northwest Rotorcraft (R00005SE)

Part Number 204P2100-101 and -103 Main Rotor Blade is applicable on the following UH-1B Models with STC SR00026DE Installed:

Rotorcraft Dev. Corp. (H3NM) Rotorcraft Dev. Corp. (H13WE) San Joaquin Helicopters (H1RM) International Helicopters (H5SO) Richards Heavylift Helo, Inc. (H3SO) OAS Parts LLC (H7SO)

Note: On UH-1B Models with STC SR00026DE Installed, reference FAA STC Number SR00026DE and San Joaquin Helicopters Helicopter Maintenance Manual Supplement No. 1 to Army Model UH-1B Helicopter (Restricted Category).

Part Number 204P2100-101 and -103 Main Rotor Blade is applicable on the following TH-1F Models:

Rotorcraft Dev. Corp. (H12NM) Tamarack Helicopters (H7NE) Robinson Air Crane, Inc. (R00008AT)

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HELICOPTER TECHNOLOGY COMPANY MAINTENANCE MANUAL HTCM-006 UH-1H MAIN ROTOR BLADE Change Letter: D Change Date: 01/03/2020

Part Number 204P2100-101 and -103 Main Rotor Blade is applicable on the following UH-1F Models:

Rotorcraft Dev. Corp. (H12NM) Tamarack Helicopters (H7NE) Robinson Air Crane, Inc. (R00008AT) AST, Inc (H11SW) California Department of Forestry (H2NM)

Part Number 204P2100-101 and -103 Main Rotor Blade is applicable on the following UH-1P Models:

Rotorcraft Dev. Corp. (H12NM) Robinson Air Crane, Inc. (R00008AT)

Part Number 204P2100-101 and -103 Main Rotor Blade is applicable on the following 204B Model:

Bell Helicopter Textron, Inc. (H1SW)

Part Number 204P2100-101 and -103 Main Rotor Blade is applicable on the following 205A Model:

Bell Helicopter Textron, Inc. (H1SW)

Part Number 204P2100-101101 and -103 Main Rotor Blade is applicable on the following 205A-1 Model:

Bell Helicopter Textron, Inc. (H1SW)

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 HELICOPTER TECHNOLOGY COMPANY MAINTENANCE MANUAL

 HTCM-006 UH-1H MAIN ROTOR BLADE

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204P2100-101 AND -103 MAIN ROTOR BLADE ASSEMBLY:

FOR UH-1H MODELS (TM 55-1520-210-23-1) AND FOR UH-1B MODELS WITH STC NUMBER SR00026DE INSTALLED (TM 55-1520-210-23-1)

CHAPTER 1. INTRODUCTION

Section I. Servicing

1-7. Main and Tail Rotor System

Refer to Paragraph 1-7.

- 1-14. Cleaning
- 1-15. Description Cleaning

Refer to Paragraph 1-15.

1-19. Rotor Blades - Cleaning

Refer to Paragraph 1-19.

1-20. Treatment of Aluminum and Magnesium Corrosion

Refer to Paragraph 1-20.

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1-21. Snow and Ice Removal

Refer to Paragraph 1-21.

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- 1-22. Consumable Maintenance Supplies and Materials
- 1-23. Description Consumable Maintenance Supplies and Materials

Refer to Paragraph 1-23.

Table 1-1. Consumable Maintenance Supplies and Materials

Refer to Table 1-1.

- 1-24. Special Tools and Test Equipment
- 1-25. Description Special Tools and Test Equipment

Refer to Paragraph 1-25.

Table 1-2. Special Tools and Test Equipment

Refer to Table 1-2.

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HELICOPTER TECHNOLOGY COMPANY MAINTENANCE MANUAL HTCM-006 UH-1H MAIN ROTOR BLADE Change Letter: D Change Date: 01/03/2020

- 1-26. Support Equipment
- 1-27. Description Support Equipment

Refer to Paragraph 1-27.

Table 1-3. Support Equipment

Refer to Table 1-3.

1-28. Standard Torque Procedures and Requirements

Refer to Paragraph 1-28.

Table 1-4. Standard Torque Chart

Refer to Table 1-4.

1-29. Reuse of Self-Locking Nuts

Refer to Paragraph 1-29.

Table 1-7. Minimum Breakaway Torque

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HELICOPTER TECHNOLOGY COMPANY MAINTENANCE MANUAL

HTCM-006 UH-1H MAIN ROTOR BLADE

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SECTION IV. INSPECTION REQUIREMENTS

1-56. General Information

Refer to Paragraph 1-56.

1-57. Standards of Serviceability

Refer to Paragraph 1-57.

- 1-58. Special Inspection
- 1-59. Description Special Inspection

Refer to Paragraph 1-59.

1-60. Definition and General Information - Special Inspection

Refer to Paragraph 1-60.

1-61. Requirements - Special Inspection

Refer to Paragraph 1-61.

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Section V. Overhaul and Retirement Schedule

1-62. Introduction

Refer to Paragraph 1-62.

- 1-63. Overhaul Interval
- 1-64. Description Overhaul Interval

Refer to Paragraph 1-64.

- 1-65. Retirement Schedule
- 1-66. Description Retirement Schedule

Refer to Paragraph 1-66.

Table 1-8. Overhaul and Retirement Schedule

Refer to Table 1-8.

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Section VI. Flight Safety Critical Aircraft Parts

1-67. Flight Safety Critical Aircraft Parts

Refer to Paragraph 1-67.

Table 1-9. Flight Safety Critical Aircraft Parts

Refer to Table 1-9.

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CHAPTER 5. ROTORS

Section I. Main Rotor System

- 5-1. Main Rotor System
- 5-2. Description Main Rotor System

Refer to Paragraph 5-2.

- 5-3. Main Rotor Hub and Blade Assembly
- 5-4. Description Main Rotor Hub and Blade Assembly

Refer to Paragraph 5-4.

5-5. Cleaning - Main Rotor Hub and Blade Assembly

Refer to Paragraph 5-5.

5-6. Lubrication - Main Rotor Hub and Blade Assembly

Refer to Paragraph 5-6.

5-7. Alignment - Main Rotor Hub and Blade Assembly

Refer to Paragraph 5-7.

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5-8. Adjustment, Collective Pitch Forces - Main Rotor Hub with Metal Blade Installed

Refer to Paragraph 5-8.

5-9. Operational Check - Main Rotor Hub and Blade Assembly

Refer to Paragraph 5-9.

5-10. Autorotation RPM Adjusting - Main Rotor Hub and Blade Assembly

Refer to Paragraph 5-10.

5-11. Troubleshooting - Main Rotor Hub and Blade Assembly

Refer to Paragraph 5-11.

Table 5-1. Troubleshooting Main Rotor System

Refer to Table 5-1.

5-12. Removal - Main Rotor Hub and Blade Assembly

Refer to Paragraph 5-12.

5-13. Installation - Main Rotor Hub and Blade Assembly

Refer to Paragraph 5-13.

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Section III. Main Rotor Blades

- 5-27. Main Rotor Blade
- 5-28. Description Main Rotor Blade

Refer to Paragraph 5-28.

5-29. Inspection - Main Rotor Blade (Installed)

Refer to Paragraph 5-29.

Table 5-3. Inspection Requirements Main Rotor Blade

Refer to Table 5-3.

5-30. Removal - Main Rotor Blade

Refer to Paragraph 5-30.

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5-31. Inspection - Main Rotor Blade

Refer to Paragraph 5-31.

Table 5-4. Main Rotor Blade - Repairable Nicks, Scratches, and Corrosion Limits

Refer to Table 5-4.

Table 5-5. Main Rotor Blade - Dent Limits

Refer to Table 5-5.

5-32. Repair or Replacement - Main Rotor Blade

Refer to Paragraph 5-32.

5-33. Installation - Main Rotor Blade

Refer to Paragraph 5-33.

5-34. Touchup Refinish Procedure - Main Rotor Blade

Refer to Paragraph 5-34.

5-34.1. Preparation for Storage or Shipment - Main Rotor Blade

Refer to Paragraph 5-34.1

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Section IX. Tracking Procedures

5-120. Tracking and Operational Check - Main Rotor Blades

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5-121. Vibration Analysis - Main Rotor Blades

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5-122. Deleted

Paragraph 5-122 has been Deleted.

5-125. Tracking and Balancing with the Vibrex 4591 System

5-126. General

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5-127. Equipment Description

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5-128. Main Rotor Track and Balance

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5-129. Attach Test Equipment to Aircraft

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5-130. Hover Track of Main Rotor with Metal Blades

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5-131. Check Main Rotor Balance

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5-132. Correct Main Rotor Balance

Refer to Paragraph 5-132.

5-133. Check Main Rotor In-Flight Track

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5-134. Correct In-Flight Track

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5-140. Vibration Source Location

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Table 5-7. Operating Speeds

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5-141. Troubleshooting

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204P2100-101 AND -103 MAIN ROTOR BLADE ASSEMBLY: FOR TH-1F MODELS (T.O. 1H-1(U)F-2-1), FOR UH-1F MODELS (T.O. 1H-1(U)F-2-1), AND FOR UH-1P MODELS (T.O. 1H-1(U)F-2-1)

I. GENERAL INFORMATION

1-1. General Information

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1-13. Leading Particulars

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Table 1-1. Leading Particulars

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1-14. Station Locations

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1-31. Drive System – Preservation (Inoperable Engine)

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1-40. Depreservation

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1-52. Main Rotor Assembly

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1-53. Extreme Climactic Environmental Maintenance

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1-54. Extremes in Temperature

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2-2. General

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2-3. Towing

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2-5. Towing

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Figure 2-1. Towing and Parking

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2-34. Parking

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2-35.

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Parking Procedures

2-36. Anchoring and Mooring

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2-37. Mooring Procedures

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Figure 2-14. Typical Mooring

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2-38. Blade Tie Down

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2-39. Jacking

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2-41. Jacking Procedure

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2-43. Jacking for Weighing

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2-48. Maintenance Hoist

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2-56. Cold Weather Ground Check

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2-58. General

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2-71. General

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2-74. Torque Requirements

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2-75. Torqueing Instructions

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2-76. Torqueing Limits

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2-80. Measuring Effective Length of Crowfoot Wrench

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2-81. Determination of Gage Reading when using a Crowfoot Wrench

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Figure 2-30. Torque Application using Extension Wrench

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2-87. Special Tools and Equipment

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Table 2-4. Special Tools and Equipment

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Figure 7-2. Rotor System Torque Values

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7-4. Alternate Removal – Main Rotor Hub and Blade Assembly

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7-4A. Removal of Pitch Change Link Assembly

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7-5. Removal – Main Rotor Blades

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7-6. Inspection and Repair Main Rotor Blades

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7-24. Clean Main Rotor Blades

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7-31. Tracking and Balancing Main and Tail Rotor Assembly

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7-32. Vibrex System Tracking and Balancing Main and Tail Rotor Blades and Hubs

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7-33. Description of Equipment

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Figure 7-27. Balance Chart Clock Angle Corrector

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Figure 7-28. Reclocked Chart

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7-35. Installation of Equipment

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Figure 7-31. Installation of Interrupter

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Figure 7-32. Adjustment of Magnetic Pickup Clearance

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Refer to Figure 7-33.

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Figure 7-34. Installation of Retro-Reflective Tape to Main Rotor Tip Targets

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Figure 7-35. Installation of Tip Targets

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Figure 7-36. Installation of Accelerometer and Bracket, PN 3382

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Figure 7-40. Hover and Ground Track Blade Pattern

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7-37. Dynamic Balancing of Main Rotor

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7-38. Inflight Tracking Main Rotor Blades

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7-43. Tracking Flag Method

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7-44. Vibration Check and Adjustment of Main Rotor

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Figure 7-45. Main Rotor Tracking Procedure

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Figure 7-46. Rotor Smoothing Procedure

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7-45. Sweeping Blade of Main Rotor

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7-46. Autorotation RPM Adjustment of Main Rotor

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Figure 7-47. Troubleshooting Lateral Vibration

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Figure 7-48. Trim Tab Bender and Gage Application

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Figure 7-49. Tracking Flag

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7-47. Adjusting for Collective Forces

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7-48. Adjustment - Collective Pitch Force – Main Rotor Hub and Blade Assembly

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Figure 7-50. Collective Pitch Retention Strap Adjustment

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7-49. Resetting Tension Torsion Straps to Initial Setting

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7-50. Resetting Main Rotor Grip Strap

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7-51. Troubleshooting Rotors and Controls

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Table 7-2. Troubleshooting Rotors and Controls

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7-101. Packaging and Preservation of Components

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7-102. Installing Main Rotor Blades in Shipping Containers

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204P2100-101 AND -103 MAIN ROTOR BLADE ASSEMBLY: FOR 204B MODELS (BHT-204B-M&O-1)

INTRODUCTION

1. Use of the Manual

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2. Bulletins

Refer to Paragraph 2.

3. Consumable Materials

Refer to Paragraph 3.

4. Special Tools

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5. Torques

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6. Terminology

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7. Warnings, Cautions, and Notes

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8. Use of Procedural Words

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9. Wear Limits

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10. Standard Practices

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11. Replacement Parts and Assemblies

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Figure 1-1. 204B Helicopter

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1-1. General Information

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1-2. Description

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1-3. General

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1-7. Main Rotor

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1-20. Helicopter Dimensions

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1-21. Ground Handling

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Figure 1-2. Three-View Dimensional Diagram

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Figure 1-3. Station Line Diagram

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Figure 1-4. Ground Handling (Hoisting, Jacking, Leveling, and Towing)

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1-29. Parking - Normal Conditions

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1-30. Parking – Turbulent Conditions

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1-31. Mooring

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Figure 1-5. Parking and Mooring

Refer to Figure 1-5.

1-32. Helicopter Storage

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1-33. Environmental Conditions

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1-34. Flyable Storage

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1-35. Short Term Storage

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1-36. Intermediate Storage

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1-67. Maintenance Hoist

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1-78. Inspection

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1-79. Daily Inspection

Refer to Paragraph 1-79.

1-79. Daily Inspection

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1-80. 100 Hour Inspection

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1-82. 3000 Hour Inspection

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1-83. 1000 Hour Component Overhaul

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1-85. 2000 Hour Component Overhaul

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1-86. 2400 Hour Component Overhaul

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1-86A. 2500 Hour Component Overhaul

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1-87. Between 5 and 10 Hours of Flight After Installation

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1-88. Each 10 Hours of Component Operation

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1-89. Each 25 Hours of Component Operation

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1-90. Between 25 and 30 Hours of Flight After Installation

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1-91. Each 50 Hours or 15/30 Days

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1-92. 100 Hours After Initial Installation of Tailboom

Refer to Paragraph 1-92.

1-93. Each 100 Hours

Refer to Paragraph 1-93.

1-94. Each 100 Hours or 3 Months, Whichever Occurs First

Refer to Paragraph 1-94.

1-95. Each 300 Hours or 3 Months, Whichever Occurs First

Refer to Paragraph 1-95.

1-96. Each 500 Hours or 12 Months, Whichever Occurs First

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1-96A. Each 24 Months

Refer to Paragraph 1-96A.

1-96B. Deleted

Paragraph 1-96B has been Deleted.

1-97. Each 1200 Hours or 24 Months, Whichever Occurs First

Refer to Paragraph 1-97.

1-98. Hard Landing

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1-99. Sudden Stoppage – Power On or Off

Refer to Paragraph 1-99.

1-100. Overspeed

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1-102. Compressor Surge or Stall

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1-104. Magnetic Compass Malfunction

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1-105. Overhaul Evaluation Criteria

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SECTION II. MAIN ROTOR HUB AND BLADE ASSEMBLY

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2-2. Description

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2-3. Removal – Main Rotor Hub and Blade Assembly

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2-4. Main Rotor Blades

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2-5. Description

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2-6. Removal – Main Rotor Blades

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Refer to Figure 2-1.

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2-7. Installation – Main Rotor Blades

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Figure 2-3. Removing Main Rotor Blades

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Figure 2-4. Main Rotor Blade Retention Bolt Extracting Fixture

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2-9. Leading Edge Protection – Main Rotor Blades (Polyurethane Tape)

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2-10. Daily Inspection – Main Rotor Blades

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2-11. 1000 Hour Inspection and Repair – Main Rotor Blades

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Figure 2-5. Main Rotor Blade Repair

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2-12. Conditional Inspection – Main Rotor Blades

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2-13. Inspection – Main Rotor Blades

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2-14. Field Repairs – Main Rotor Blades

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2-15. Trim Tab Replacement – Main Rotor Blade

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2-16. Main Rotor Blade Splice Cover

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2-17. Installation – Main Rotor Blade Splice Cover

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2-18. Preservation, Storage, and Blade Packaging

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2-19. Main Rotor Hub

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2-43. Balancing - Main Rotor Hub Assembly

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Figure 2-34. Balancing Main Rotor Hub

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2-44. Installation – Main Rotor Blades

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2-45. Alignment – Main Rotor Blades

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2-46. Installation – Main Rotor Hub and Blade Assembly

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Figure 2-35. Main Rotor Blade Alignment

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2-47. Torque Limits – Main Rotor

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2-48. Maintenance Operational Check - Main Rotor

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2-49. Tracking – Main Rotor Blades

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Figure 2-36. Rotor System Torque Limits

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Figure 2-37. Tracking Main Rotor Blades

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2-50. Spanwise Balance Check

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2-51. Chordwise Balance Check

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2-52. Autorotation RPM Check

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2-53. Collective Pitch Force Check and Adjustment

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2-54. Main Rotor Hub Sealing

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204P2100-101 AND -103 MAIN ROTOR BLADE ASSEMBLY: FOR 205A MODELS (BHT-205A1-MM-1) AND FOR 205A-1 MODELS (BHT-205A1-MM-1)

CHAPTER 1. INTRODUCTION

1-1. General

Refer to Paragraph 1-1.

1-2. Helicopter Description

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Figure 1-1. Model 204A-1 Helicopter (Typical)

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1-3. Use of the Manual

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1-4. Bulletins

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1-5. Consumable Materials

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Refer to Paragraph 1-5.

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1-6. Special Tools

Refer to Paragraph 1-6.

1-7. Torques

Refer to Paragraph 1-7.

1-8. Terminology

Refer to Paragraph 1-8.

1-9. Warnings, Cautions, and Notes

Refer to Paragraph 1-9.

1-10. Use of Procedural Words

Refer to Paragraph 1-10.

1-11. Wear Limits

Refer to Paragraph 1-11.

1-12. Standard Practices

Refer to Paragraph 1-12.

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1-13. Replacement Parts and Assemblies

Refer to Paragraph 1-13.

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CHAPTER 4. AIRWORTHINESS LIMITATIONS

4-1. Airworthiness Limitations Schedule

Refer to Paragraph 4-1.

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CHAPTER 5. INSPECTIONS

Inspections

- 5-1. General
 - Refer to Paragraph 5-1.
- 5-2. Inspection Requirements
 - Refer to Paragraph 5-2.
- 5-3. Crash Damage
 - Refer to Paragraph 5-3.
- 5-4. Types of Inspections
 - Refer to Paragraph 5-4.
- 5-5. Definitions
 - Refer to Paragraph 5-5.
- 5-6. Inspection and Overhaul Tolerance

Refer to Paragraph 5-6.

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Scheduled Inspections

5-7. Scheduled Inspections

Refer to Paragraph 5-7.

5-8. Daily Inspection

Refer to Paragraph 5-8.

5-9. 100 Hour Inspection

Refer to Paragraph 5-9.

5-10. 1000 Hour Inspection

Refer to Paragraph 5-10.

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Special Inspections

5-11. Special Inspections

Refer to Paragraph 5-11.

5-12. Daily or Each 10 Hours of Flight Operation, Whichever Occurs First until 250 Hours

Refer to Paragraph 5-12.

5-13. Between 5 and 10 Hours of Flight After Installation

Refer to Paragraph 5-13.

5-14. Each 8 Hours of Component Operation

Refer to Paragraph 5-14.

5-15. Each 25 Hours of Component Operation

Refer to Paragraph 5-15.

5-16. Each 25 Hours for the Next Four Inspections

Refer to Paragraph 5-16.

5-17. Each 25 Hours of Component Operation or 15 Days, or Each 5 Days for Blades Operating in Salt Laden Atmosphere, Whichever Occurs First

Refer to Paragraph 5-17.

5-18. 100 Hours After Each Installation

Refer to Paragraph 5-18.

5-19. Main Rotor Grip Ultrasonic Inspection

Refer to Paragraph 5-19.

5-20. Each 300 Hours of Component Operation

Refer to Paragraph 5-20.

5-21. Each 300 Hours or 3 Months of Component Operation

Refer to Paragraph 5-21.

5-22. Each 500 Hours of Component Operation

Refer to Paragraph 5-22.

5-23. Each 600 Hours/6 Months of Tail Rotor Driveshaft Coupling Operation

Refer to Paragraph 5-23.

5-24. Each 600 Hours or 12 Months of Component Operation

Refer to Paragraph 5-24.

5-25. Each 6 Months

Refer to Paragraph 5-25.

5-26. Each 12 Months

Refer to Paragraph 5-26.

5-27. Each 1000 Hours or 12 Months of Component Operation

Refer to Paragraph 5-27.

5-28. First 1000 Hours of Component Time and Each 3000 Hours Thereafter of Component Time

Refer to Paragraph 5-28.

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5-29. Each 1200 Hours of Component Operation - Deleted

Paragraph 5-29 has been Deleted.

5-30. Each 1200 Hours or 24 Months, Whichever Comes First

Refer to Paragraph 5-30.

5-31. Each 24 Months of Flight Control System Bolt Operation

Refer to Paragraph 5-31.

5-32. Each 3000 Hours of Component Operation

Refer to Paragraph 5-32.

5-33. Each 3100 Hours of Component Operation

Refer to Paragraph 5-33.

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Conditional Inspections

5-34. Conditional Inspections

Refer to Paragraph 5-34.

5-35. Hard Landing

Refer to Paragraph 5-35.

5-36. After Blade Strike or Other Rotating System Torque Spike

Refer to Paragraph 5-36.

5-37. Sudden Stoppage/Acceleration – Main Rotor

Refer to Paragraph 5-37.

5-38. Sudden Stoppage/Acceleration – Tail Rotor

Refer to Paragraph 5-38.

5-39. Overspeed

Refer to Paragraph 5-39.

5-40. Overtorque

Refer to Paragraph 5-40.

5-41. Compressor Stall or Surge

Refer to Paragraph 5-41.

5-42. Lightning Strikes

Refer to Paragraph 5-42.

5-43. Magnetic Compass Malfunction

Refer to Paragraph 5-43.

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Component Overhaul Schedule

5-44. Component Overhaul Schedule

Refer to Paragraph 5-44.

Table 5-1. Component Overhaul Schedule

Refer to Table 5-1.

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CHAPTER 6. DIMENSIONS AND CHARTS

6-1. Principal Dimensions

Refer to Paragraph 6-1.

6-2. Stations, Waterlines, and Buttocklines

Refer to Paragraph 6-2.

Figure 6-1. Principal Dimensions

Refer to Figure 6-1.

Figure 6-2. Station Diagram

Refer to Figure 6-2.

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CHAPTER 7. LIFTING AND JACKING

7-1. Lifting and Jacking

Refer to Paragraph 7-1.

7-2. Lifting the Complete Helicopter

Refer to Paragraph 7-2.

7-3. Lifting the Tailboom Only

Refer to Paragraph 7-3.

7-4. Jacking

Refer to Paragraph 7-4.

Figure 7-1. Jacking

Refer to Figure 7-1.

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CHAPTER 9. TOWING

9-1. Towing

Refer to Paragraph 9-1.

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CHAPTER 10. PARKING AND MOORING

10-1. Parking – Normal Conditions

Refer to Paragraph 10-1.

10-2. Parking – Turbulent Conditions

Refer to Paragraph 10-2.

10-3. Mooring

Refer to Paragraph 10-3.

Figure 10-1. Parking and Mooring

Refer to Figure 10-1.

10-4. Helicopter Storage

Refer to Paragraph 10-4.

10-5. Environmental Conditions

Refer to Paragraph 10-5.

10-6. Flyable Storage

Refer to Paragraph 10-6.

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10-7. Short Term Storage

Refer to Paragraph 10-7.

10-8. Intermediate Storage

Refer to Paragraph 10-8.

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CHAPTER 65. ROTOR SYSTEM

65-1. Rotor System

Refer to Paragraph 65-1.

65-2. Vibration Analysis

Refer to Paragraph 65-2.

65-3. Extreme Low Frequency Vibration

Refer to Paragraph 65-3.

65-4. Low Frequency Vibration

Refer to Paragraph 65-4.

65-5. Medium Frequency Vibration

Refer to Paragraph 65-5.

65-6. High Frequency Vibration

Refer to Paragraph 65-6.

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65-7. Main Rotor Troubleshooting

Refer to Paragraph 65-7.

Table 65-1. Main Rotor Troubleshooting

Refer to Table 65-1.

65-9. Operational Check - Main Rotor System

Refer to Paragraph 65-9.

65-10. Main Rotor Tracking

Refer to Paragraph 65-10.

Figure 65-1. Main Rotor Tracking Procedure

Refer to Figure 65-1.

65-11. Main Rotor Blade Vibration Check and Adjustment

Refer to Paragraph 65-11.

65-12. Main Rotor Blade Sweeping

Refer to Paragraph 65-12.

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65-13. Main Rotor Autorotation RPM Adjustment

Refer to Paragraph 65-13.

Figure 65-2. Lateral Vibration Check

Refer to Figure 65-2.

Figure 65-3. Rotor Smoothing Procedure

Refer to Figure 65-3.

Figure 65-4. Trim Tab Bender and Gage

Refer to Figure 65-4.

65-14. Main Rotor Hub and Blade

Refer to Paragraph 65-14.

65-15. Removal – Main Rotor Hub and Blade

Refer to Paragraph 65-15.

65-15A. Inspection and Repair – Main Rotor Hub and Blade

Refer to Paragraph 65-15A.
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65-16. Installation – Main Rotor Hub and Blade

Refer to Paragraph 65-16.

Figure 65-5. Main Rotor System

Refer to Figure 65-5.

Figure 65-6. Main Rotor Retaining Nut Damage and Repair Limits

Refer to Figure 65-6.

Figure 65-7. Main Rotor Cone Set Damage and Repair Limits

Refer to Figure 65-7.

65-17. Minimum Blade Angle – Main Rotor Hub and Blade

Refer to Paragraph 65-17.

65-18. Collective Pitch Forces – Adjustment

Refer to Paragraph 65-18.

Figure 65-8. Main Rotor System Torque Values

Refer to Figure 65-8.

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Figure 65-9. Collective Pitch Retention Strap Adjustment

Refer to Figure 65-9.

65-19. Resetting Main Rotor Grip Tension-Torsion Strap

Refer to Paragraph 65-19.

65-20. Resetting Main Rotor Grip Tension-Torsion Strap – Initial Setting

Refer to Paragraph 65-20.

65-21. Main Rotor Blades

Refer to Paragraph 65-21.

65-22. Preventative Maintenance for Main Rotor Blades

Refer to Paragraph 65-22.

65-23. Main Rotor Blade Daily Inspection

Refer to Paragraph 65-23.

65-24. Nonrepairable Damage - Main Rotor Blades

Refer to Paragraph 65-24.

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65-25. Inspection and Minor Repair - Main Rotor Blades

Refer to Paragraph 65-25.

Figure 65-10. Main Rotor Blade

Refer to Figure 65-10.

65-26. Field Repairs - Main Rotor Blades

Refer to Paragraph 65-26.

Figure 65-11. Main Rotor Blade Repair

Refer to Figure 65-11.

65-27. Polyurethane Tape - Installation

Refer to Paragraph 65-27.

65-28. Deleted

Paragraph 65-28 has been Deleted.

65-29. Main Rotor Blade Paint Touch-Up

Refer to Paragraph 65-29.

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65-30. Main Rotor Blade Trim Tab

Refer to Paragraph 65-30.

65-31. Replacement - Main Rotor Blade - Trim Tab

Refer to Paragraph 65-31.

Figure 65-12. Trim Tab Replacement

Refer to Figure 65-12.

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HTC SPECIFIC INSPECTIONS

For 204P2100-101 Only: Main Rotor Blade Root End Periodic Inspection Requirements

REQUIRED MATERIALS

- 1. Naphtha or Rubbing Alcohol
- 2. Kim Wipes or equivalent clean lint-free wipes.
- 3. Bright Flashlight
- 4. 10x Magnifying Glass

INSPECTION INSTRUCTIONS

Within 25 Hours time-in-service (TIS) or 2 weeks, whichever occurs first, and thereafter at intervals not to exceed 25 hours TIS or 2 weeks, whichever occurs first, perform the following:

- Clean the upper and lower exposed surfaces of each M/R Blade from an area starting at the Butt (Inboard-most) End of the Blade to three (3.0) inches outboard of the Doublers.
- Use a 10X Magnifying Glass and Flashlight, visually inspect the M/R blade parts for a crack or corrosion.
- If there is a crack, corrosion, an edge void, loose or damaged adhesive squeeze-out, or an edge delamination, the Blade is Not Airworthy and must be replaced with an Airworthy Blade before further flight.

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- Record compliance with these inspections in the Rotorcraft Log Book and/or in the Technical Directives and Bulletins section of the rotor blade Serviceable Component Record, as applicable.
- For further information and rotor blade disposition, contact Helicopter Technology Company, LLC (HTC) at (310) 523-2750, or FAX (310) 523-2745.

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