

SUPPLEMENTAL TYPE CERTIFICATE

10073531

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

HELICOPTER TECHNOLOGY COMPANY

12902 SOUTH BROADWAY
LOS ANGELES CA 90061
UNITED STATES OF AMERICA

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: US H1SW

Type Certificate Holder: BELL TEXTRON, INC.

Type: Bell 204/205/210

Model: 204B, 205A-1

Original STC Number: FAA STC SR02492LA, 07.10.2020

Description of Design Change:

Installation, operation, and maintenance of Helicopter Technology Company main rotor blade assembly

EASA Certification Basis:

The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC SR02492LA.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

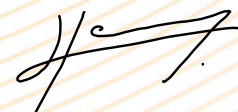
Associated Technical Documentation:

- Master Document List, MDL-204-2100-01, Rev. C, dated 24 January 2018
 - Instructions for Continued Airworthiness of Maintenance Manual HTCM-006, Rev. D, dated 03 January 2020, with its Airworthiness Limitation Section at Rev. D dated 13 April 2020
- or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the

See Continuation Sheet(s)

For the European Union Aviation Safety Agency

Cologne, Germany, 06 November 2020



Mathieu HENRY
Section Manager
Heavy Rotorcraft

Technical Implementation Procedures of EU/ USA Bilateral Agreement.

Limitations/Conditions:

Installation of HTC Main Rotor Blade with Part Number 204P2100-101 is not approved under this STC certificate.

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -

